

Euler-Lagrange Approach to Modelling

Greg and Mario

- 1 Euler-Lagrange Equations
- 2 Modelling the Rotation Dynamics (Gros2012f,Gros2013b)
- 3 Baumgarte Stabilisation (Gros2012f)
- 4 Tether Models (Pesce2003, Zanon2012, Zanon2013a)

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 - *“Everything Should Be Made as Simple as Possible, But Not Simpler”*, A. Einstein

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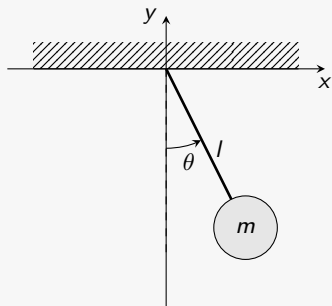
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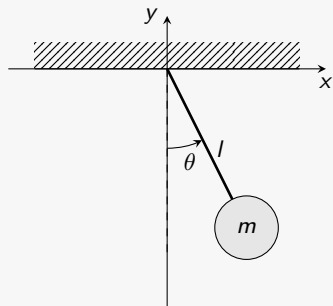
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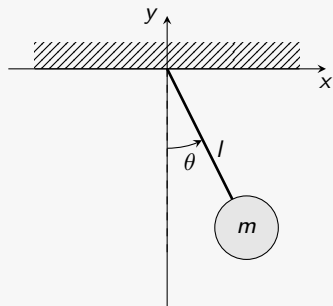
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- 6 Compute the generalised forces F^q

Example: Pendulum $q =$

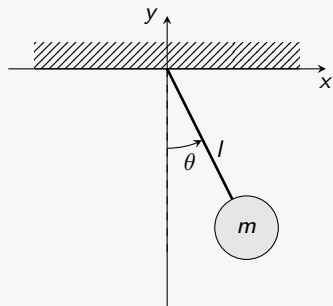
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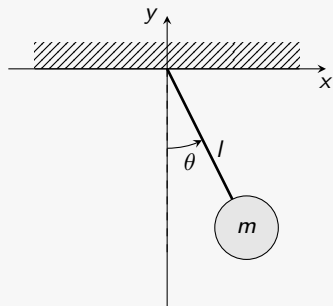
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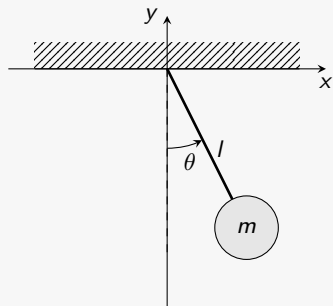
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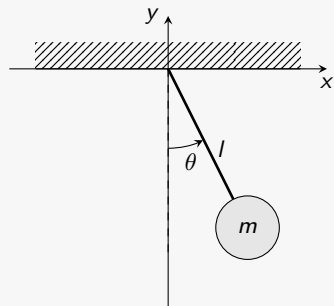
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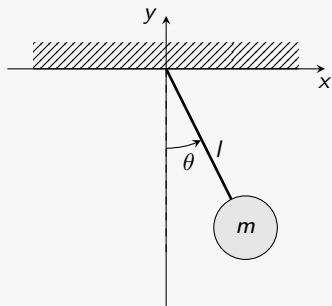
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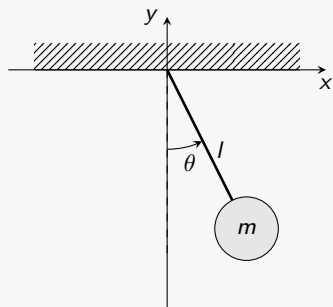
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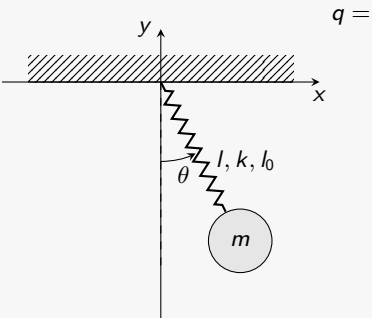
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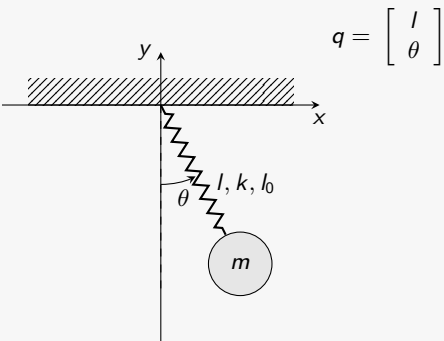
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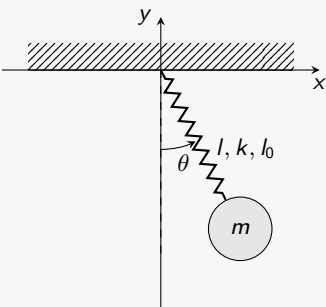
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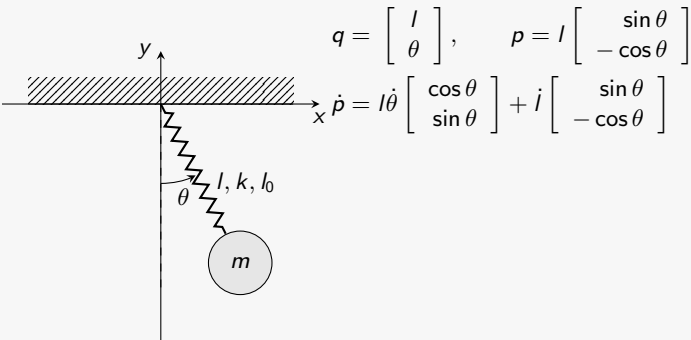
$$\ddot{\theta} = -\frac{g}{l} \sin \theta$$

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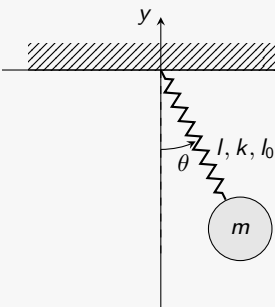
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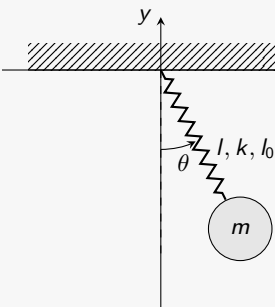


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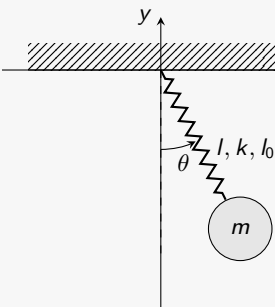


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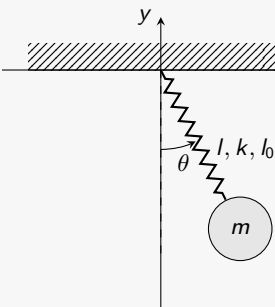
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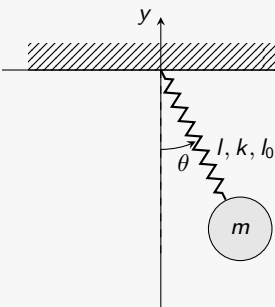
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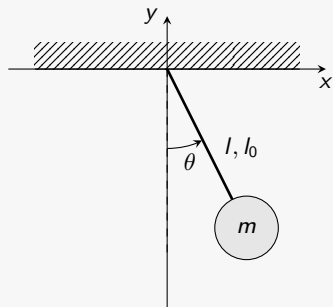
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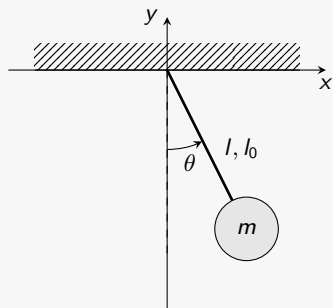
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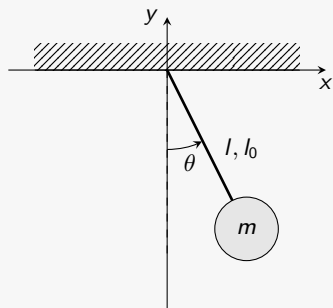
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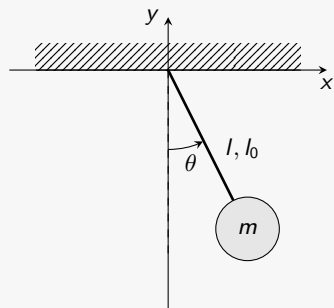
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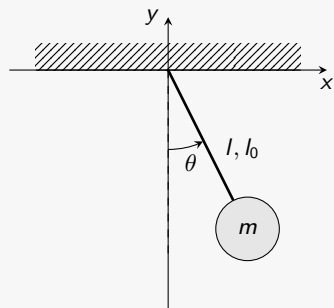
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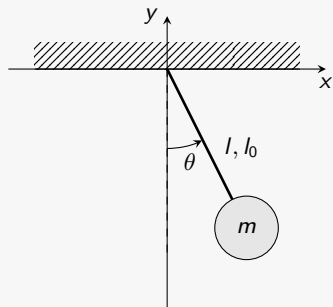
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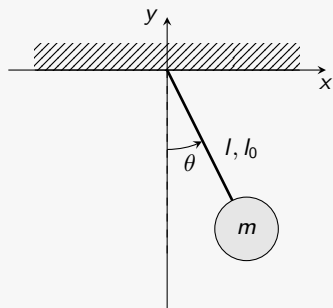
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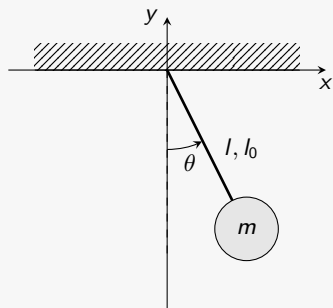
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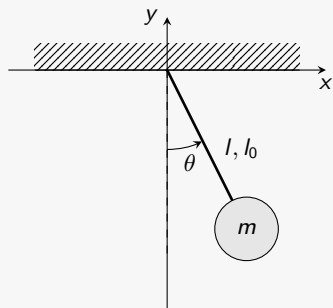
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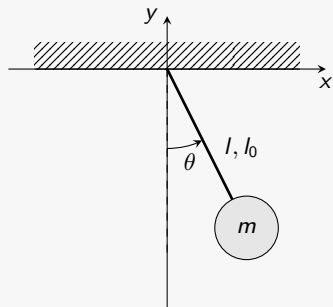
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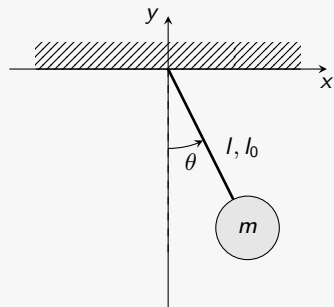
$$\frac{\partial L}{\partial q} = \begin{bmatrix} ml\dot{\theta}^2 + mg \cos \theta - \lambda \\ -mg \sin \theta \end{bmatrix}$$

$$\begin{bmatrix} \ddot{l} \\ \ddot{\theta} \end{bmatrix} = \begin{bmatrix} l\dot{\theta}^2 + g \cos \theta - \frac{\lambda}{m} \\ \frac{g}{l} \sin \theta - \frac{2}{l} \dot{\theta} \end{bmatrix}$$

Example: Constrained Pendulum

Index-3 DAE

$$\begin{bmatrix} \ddot{i} \\ \ddot{\theta} \\ 0 \end{bmatrix} = \begin{bmatrix} l\dot{\theta}^2 + g \cos \theta - \frac{\lambda}{m} \\ \frac{g}{l} \sin \theta - \frac{2}{l} \dot{\theta} \\ l - l_0 \end{bmatrix}$$

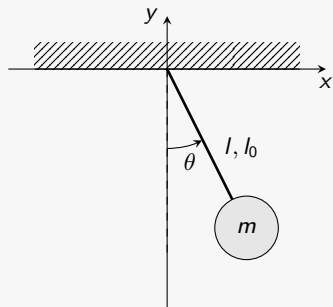
Example: Constrained Pendulum

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$$\begin{bmatrix} \ddot{i} \\ \ddot{\theta} \\ 0 \end{bmatrix} = \begin{bmatrix} l\dot{\theta}^2 + g \cos \theta - \frac{\lambda}{m} \\ \frac{g}{l} \sin \theta - \frac{2}{l} \dot{\theta} \\ l - l_0 \end{bmatrix}$$

Index reduction:

Example: Constrained Pendulum



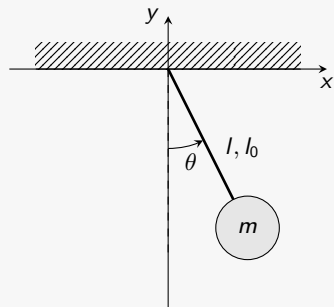
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Index reduction:

$$\dot{C} = \dot{i},$$

Example: Constrained Pendulum



Index-3 DAE

$$\begin{bmatrix} \ddot{i} \\ \ddot{\theta} \\ 0 \end{bmatrix} = \begin{bmatrix} l\dot{\theta}^2 + g \cos \theta - \frac{\lambda}{m} \\ \frac{g}{l} \sin \theta - \frac{2}{l} \dot{i}\dot{\theta} \\ l - l_0 \end{bmatrix}$$

Index reduction:

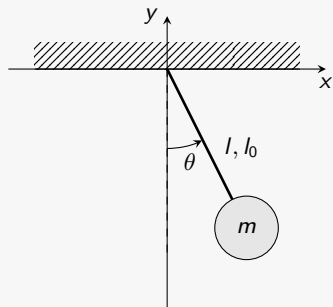
$$\dot{C} = \dot{i}, \quad \ddot{C} = \ddot{i}$$

Index-1 DAE

$$\begin{bmatrix} \ddot{i} \\ \ddot{\theta} \\ 0 \end{bmatrix} = \begin{bmatrix} l\dot{\theta}^2 + g \cos \theta - \frac{\lambda}{m} \\ \frac{g}{l} \sin \theta - \frac{2}{l} \dot{i}\dot{\theta} \\ \dot{i} \end{bmatrix}$$

$$C(t=0) = 0, \quad \dot{C}(t=0) = 0$$

Example: Constrained Pendulum



Index-3 DAE

$$\begin{bmatrix} \ddot{i} \\ \ddot{\theta} \\ 0 \end{bmatrix} = \begin{bmatrix} l\dot{\theta}^2 + g \cos \theta - \frac{\lambda}{m} \\ \frac{g}{l} \sin \theta - \frac{2}{l} \dot{i}\dot{\theta} \\ l - l_0 \end{bmatrix}$$

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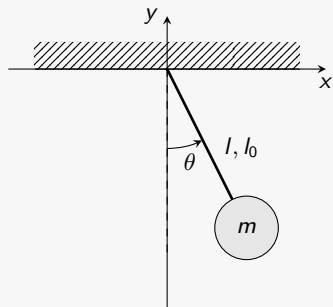
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ODE

$$\begin{bmatrix} \ddot{\theta} \\ \lambda \end{bmatrix} = \begin{bmatrix} \frac{g}{l} \sin \theta \\ ml\dot{\theta}^2 + mg \cos \theta \end{bmatrix}, \quad l = l_0$$

Example: Constrained Pendulum



$\lambda =$ tension in the rod

Index-3 DAE

$$\begin{bmatrix} \ddot{i} \\ \ddot{\theta} \\ 0 \end{bmatrix} = \begin{bmatrix} l\dot{\theta}^2 + g \cos \theta - \frac{\lambda}{m} \\ \frac{g}{l} \sin \theta - \frac{2}{l} \dot{i}\dot{\theta} \\ l - l_0 \end{bmatrix}$$

Index reduction:

$$\dot{C} = \dot{i}, \quad \ddot{C} = \ddot{i}$$

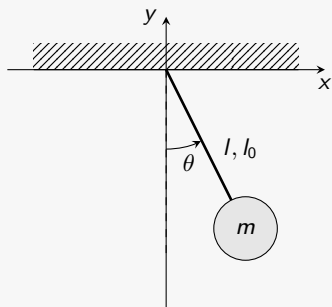
Index-1 DAE

$$\begin{bmatrix} \ddot{i} \\ \ddot{\theta} \\ 0 \end{bmatrix} = \begin{bmatrix} l\dot{\theta}^2 + g \cos \theta - \frac{\lambda}{m} \\ \frac{g}{l} \sin \theta - \frac{2}{l} \dot{i}\dot{\theta} \\ \ddot{i} \end{bmatrix}$$

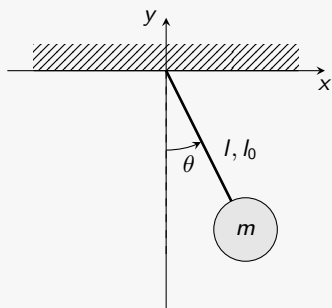
$$C(t=0) = 0, \quad \dot{C}(t=0) = 0$$

ODE

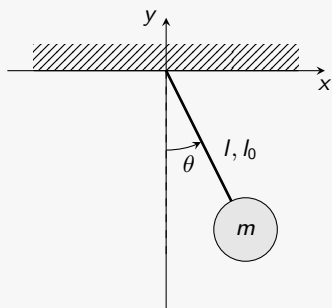
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Example: Pendulum in Cartesian Coordinates

$$q = \begin{bmatrix} x \\ y \end{bmatrix}$$

Example: Pendulum in Cartesian Coordinates

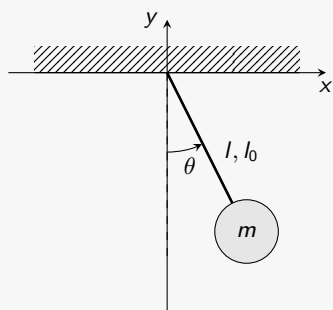
$$q = \begin{bmatrix} x \\ y \end{bmatrix}$$
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Example: Pendulum in Cartesian Coordinates

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$$\dot{p} = \begin{bmatrix} \ddot{x} \\ \ddot{y} \end{bmatrix}$$

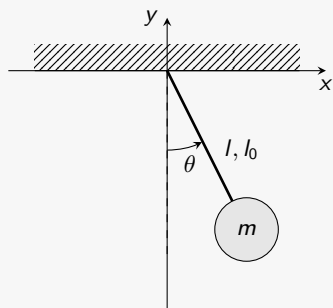
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$$T = \frac{1}{2} m \dot{p}^\top \dot{p} = \frac{1}{2} m (\dot{x}^2 + \dot{y}^2),$$

Example: Pendulum in Cartesian Coordinates

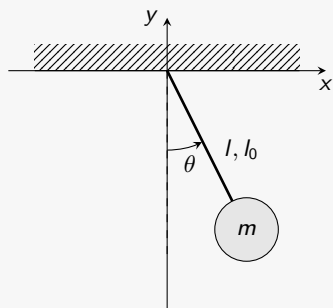


$$q = \begin{bmatrix} x \\ y \end{bmatrix}$$

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$$T = \frac{1}{2} m \dot{p}^\top \dot{p} = \frac{1}{2} m (\dot{x}^2 + \dot{y}^2), \quad U = mgy$$

Example: Pendulum in Cartesian Coordinates



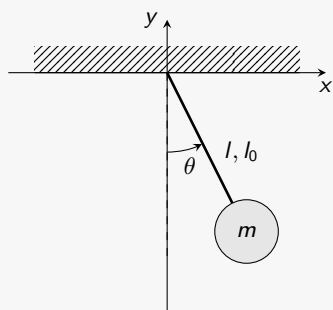
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$$C = \frac{1}{2} (x^2 + y^2 - l_0^2), \quad L = T - U - \lambda C$$

Example: Pendulum in Cartesian Coordinates



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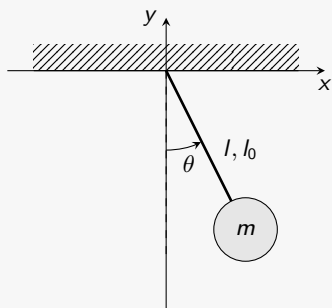
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Example: Pendulum in Cartesian Coordinates



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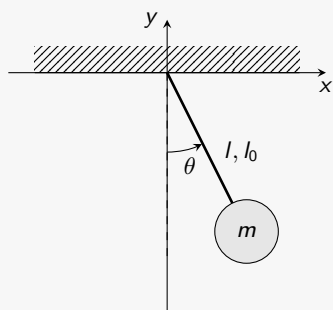
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Example: Pendulum in Cartesian Coordinates



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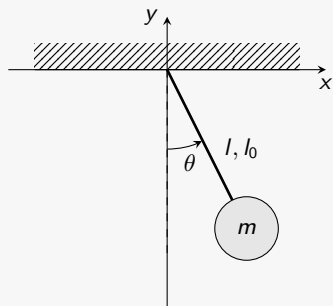
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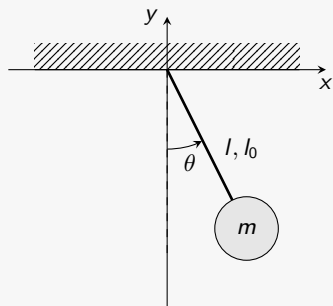
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$$\begin{bmatrix} \ddot{x} \\ \ddot{y} \\ 0 \end{bmatrix} = \begin{bmatrix} -\frac{\lambda x}{m} \\ -g - \frac{\lambda y}{m} \\ \frac{1}{2} (x^2 + y^2 - l_0^2) \end{bmatrix}$$

Example: Pendulum in Cartesian Coordinates

Index-3 DAE

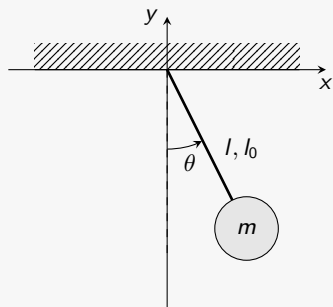
$$\begin{bmatrix} \ddot{x} \\ \ddot{y} \\ 0 \end{bmatrix} = \begin{bmatrix} -\frac{\lambda x}{m} \\ -g - \frac{\lambda y}{m} \\ \frac{1}{2}(x^2 + y^2 - l_0^2) \end{bmatrix}$$

Example: Pendulum in Cartesian Coordinates

Index-3 DAE

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Index reduction:

Example: Pendulum in Cartesian Coordinates

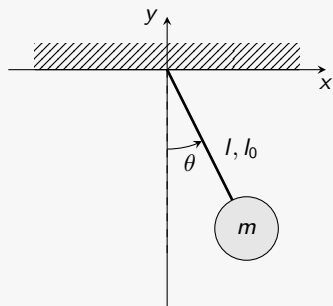
Index-3 DAE

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Index reduction:

$$\dot{C} = \dot{x}x + \dot{y}y,$$

Example: Pendulum in Cartesian Coordinates



Index-3 DAE

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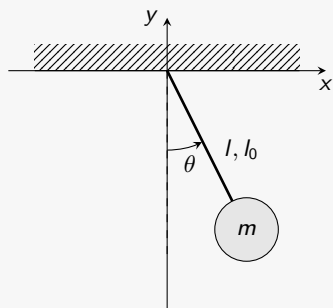
$$\dot{C} = \dot{x}x + \dot{y}y, \quad \ddot{C} = \ddot{x}x + \ddot{y}y + \dot{x}^2 + \dot{y}^2$$

Index-1 DAE

$$\begin{bmatrix} \ddot{x} \\ \ddot{y} \\ 0 \end{bmatrix} = \begin{bmatrix} -\frac{\lambda x}{m} \\ -g - \frac{\lambda y}{m} \\ \ddot{x}x + \ddot{y}y + \dot{x}^2 + \dot{y}^2 \end{bmatrix}$$

$$C(t=0) = 0, \quad \dot{C}(t=0) = 0$$

Example: Pendulum in Cartesian Coordinates



Index-3 DAE

$$\begin{bmatrix} \ddot{x} \\ \ddot{y} \\ 0 \end{bmatrix} = \begin{bmatrix} -\frac{\lambda x}{m} \\ -g - \frac{\lambda y}{m} \\ \frac{1}{2}(x^2 + y^2 - l_0^2) \end{bmatrix}$$

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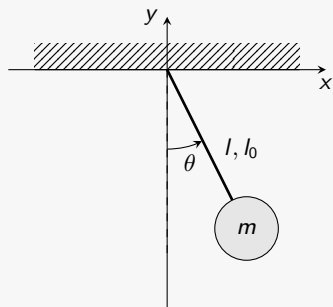
$$\begin{bmatrix} \ddot{x} \\ \ddot{y} \\ 0 \end{bmatrix} = \begin{bmatrix} -\frac{\lambda x}{m} \\ -g - \frac{\lambda y}{m} \\ \ddot{x}x + \ddot{y}y + \dot{x}^2 + \dot{y}^2 \end{bmatrix}$$

$$C(t=0) = 0, \quad \dot{C}(t=0) = 0$$

Semi-implicit form

$$\begin{bmatrix} m & 0 & x \\ 0 & m & y \\ x & y & 0 \end{bmatrix} \begin{bmatrix} \ddot{x} \\ \ddot{y} \\ \lambda \end{bmatrix} = \begin{bmatrix} 0 \\ mg \\ \dot{x}^2 + \dot{y}^2 \end{bmatrix}$$

Example: Pendulum in Cartesian Coordinates



$$\begin{bmatrix} \lambda_x \\ \lambda_y \end{bmatrix} = \text{rod tension}$$

Index-3 DAE

$$\begin{bmatrix} \ddot{x} \\ \ddot{y} \\ 0 \end{bmatrix} = \begin{bmatrix} -\frac{\lambda_x}{m} \\ -g - \frac{\lambda_y}{m} \\ \frac{1}{2}(x^2 + y^2 - l_0^2) \end{bmatrix}$$

Index reduction:

$$\dot{C} = \dot{x}x + \dot{y}y, \quad \ddot{C} = \ddot{x}x + \ddot{y}y + \dot{x}^2 + \dot{y}^2$$

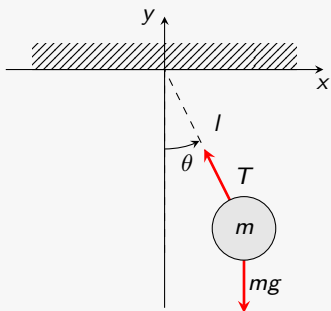
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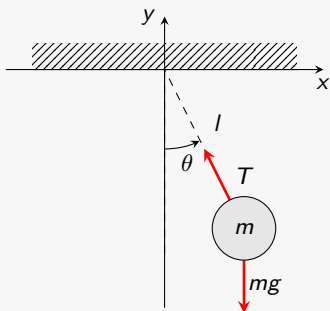
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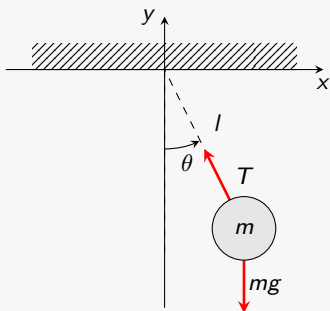
Example: Pendulum in Cartesian Coordinates**Newton's Approach**

$$m \begin{bmatrix} \ddot{x} \\ \ddot{y} \end{bmatrix} = \begin{bmatrix} T^x \\ -mg + T^y \end{bmatrix}$$

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Rod tension:

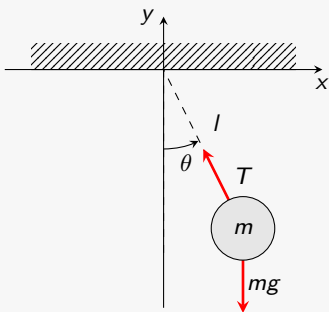
Example: Pendulum in Cartesian Coordinates**Newton's Approach**

$$m \begin{bmatrix} \ddot{x} \\ \ddot{y} \end{bmatrix} = \begin{bmatrix} T^x \\ -mg + T^y \end{bmatrix}$$

Rod tension:

$$T_g = mg \cos \theta = mg \frac{y}{\sqrt{x^2 + y^2}} = mg \frac{y}{l}$$

Example: Pendulum in Cartesian Coordinates



Newton's Approach

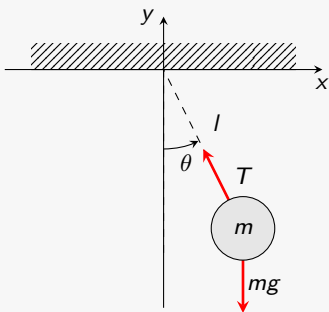
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$$T_c = ml\dot{\theta} = m \frac{\dot{x}^2 + \dot{y}^2}{l}$$

Example: Pendulum in Cartesian Coordinates



Newton's Approach

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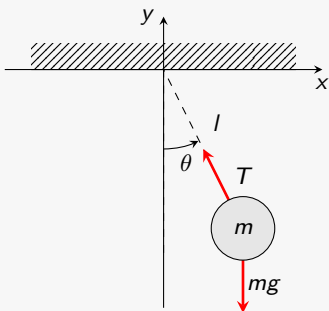
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In the x and y directions:

$$T^x = T \sin \theta = T \frac{x}{\sqrt{x^2 + y^2}} = mx \frac{\dot{x}^2 + \dot{y}^2 - gy}{l^2}$$

Example: Pendulum in Cartesian Coordinates



Newton's Approach

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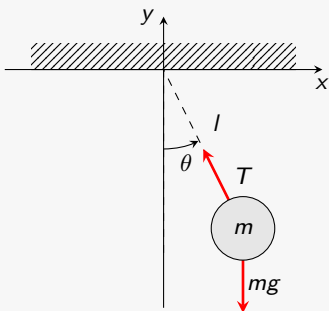
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In the x and y directions:

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$$T^y = T \cos \theta = T \frac{y}{\sqrt{x^2 + y^2}} = my \frac{\dot{x}^2 + \dot{y}^2 - gy}{l^2}$$

Example: Pendulum in Cartesian Coordinates



$$\begin{bmatrix} \ddot{x} \\ \ddot{y} \\ 0 \end{bmatrix} = \begin{bmatrix} -\frac{\lambda x}{m} \\ -g - \frac{\lambda y}{m} \\ \ddot{x}x + \ddot{y}y + \dot{x}^2 + \dot{y}^2 \end{bmatrix}$$

Newton's Approach

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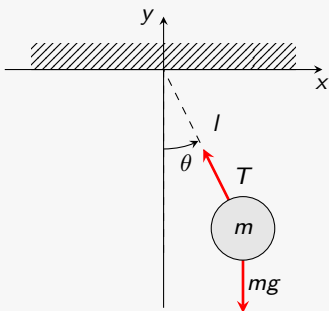
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Example: Pendulum in Cartesian Coordinates



$$\begin{bmatrix} \ddot{x} \\ \ddot{y} \\ 0 \end{bmatrix} = \begin{bmatrix} -\frac{\lambda x}{m} \\ -g - \frac{\lambda y}{m} \\ \ddot{x}x + \ddot{y}y + \dot{x}^2 + \dot{y}^2 \end{bmatrix}$$

$$-\lambda x^2 - mgy - \lambda y^2 + m\dot{x}^2 + m\dot{y}^2$$

Newton's Approach

$$m \begin{bmatrix} \ddot{x} \\ \ddot{y} \end{bmatrix} = \begin{bmatrix} T^x \\ -mg + T^y \end{bmatrix}$$

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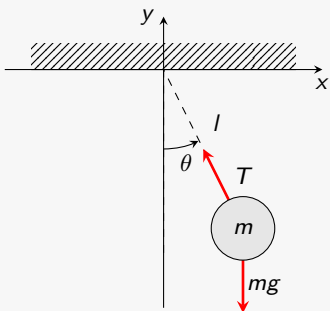
$$T_c = ml\dot{\theta} = m \frac{\dot{x}^2 + \dot{y}^2}{l}$$

In the x and y directions:

$$T^x = T \sin \theta = T \frac{x}{\sqrt{x^2 + y^2}} = mx \frac{\dot{x}^2 + \dot{y}^2 - gy}{l^2}$$

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Example: Pendulum in Cartesian Coordinates



Newton's Approach

$$m \begin{bmatrix} \ddot{x} \\ \ddot{y} \end{bmatrix} = \begin{bmatrix} T^x \\ -mg + T^y \end{bmatrix}$$

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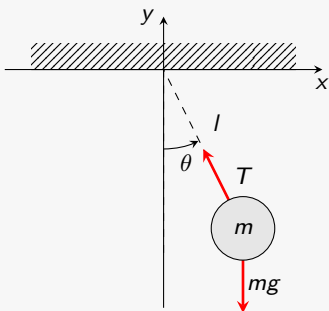
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$$-\lambda x^2 - mgy - \lambda y^2 + m\dot{x}^2 + m\dot{y}^2$$

$$\lambda = m \frac{\dot{x}^2 + \dot{y}^2 - gy}{x^2 + y^2} = m \frac{\dot{x}^2 + \dot{y}^2 - gy}{l^2}$$

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$$T^x = \lambda x, \quad T^y = \lambda y$$

Principle of Virtual Works

How to include external forces/torques?

- Generalised coordinates $q \Rightarrow$ generalised forces F^q

Typically, it is natural to express forces F in a coordinate frame $x \neq q$.

How can we compute F^q ?

- Virtual displacement: δx
- Principle of Virtual Works: $\delta W := F\delta x = F^q\delta q$
- $x = x(q)$, then: $\delta x = \frac{\partial x}{\partial q}\delta q$

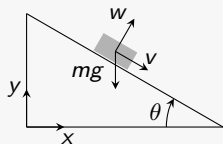
Generalised forces:

$$F^q = F \frac{\partial x}{\partial q}$$

Examples

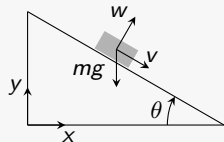
Examples

Mass on a frictionless inclined plane



Examples

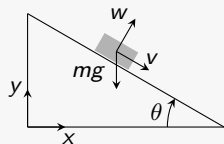
Mass on a frictionless inclined plane



$$\begin{bmatrix} x \\ y \end{bmatrix} = \begin{bmatrix} v \cos \theta + w \sin \theta \\ -v \sin \theta + w \cos \theta \end{bmatrix} = \underbrace{\begin{bmatrix} \cos \theta & \sin \theta \\ -\sin \theta & \cos \theta \end{bmatrix}}_R \begin{bmatrix} v \\ w \end{bmatrix}$$

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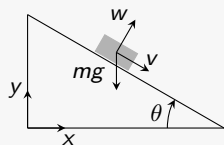


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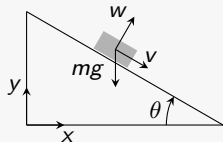


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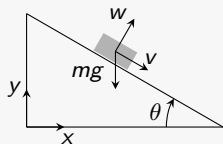


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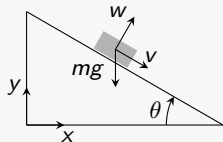
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Examples

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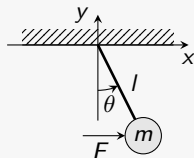


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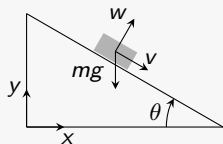
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Pendulum with Horizontal Force



Examples

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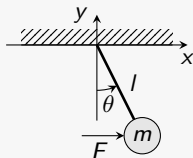


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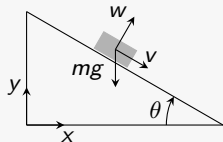
Pendulum with Horizontal Force



$$q = \theta,$$

Examples

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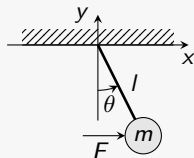


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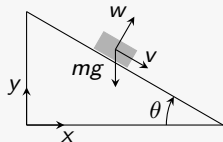
Pendulum with Horizontal Force



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Examples

Mass on a frictionless inclined plane

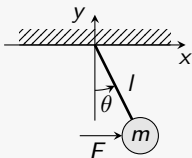


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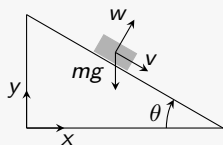
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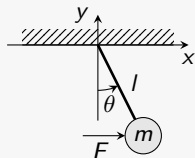


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Pendulum with Horizontal Force



$$q = \theta, \quad x = l \sin \theta, \quad F^q = F \frac{\partial x}{\partial \theta} = Fl \cos \theta$$

The generalised force F^q depends on q !

A Nice Interpretation of Euler-Lagrange

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We will use

$$\frac{\partial v}{\partial q} = \frac{d}{dt} \left(\frac{\partial x}{\partial q} \right), \quad \frac{\partial v}{\partial \dot{q}} = \frac{\partial x}{\partial q}$$

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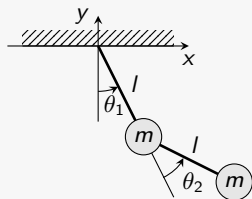
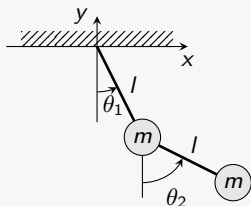
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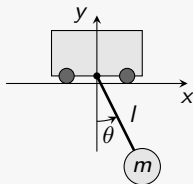
The Euler-Lagrange Equations are Newton's Equations projected on the generalised coordinates!

Exercises

- Model a double pendulum in polar and cartesian coordinates (assume $l_1 = l_2 = l$, $m_1 = m_2 = m$)



- Model a pendulum attached to a crane in polar and cartesian coordinates
- Assume to be able to change the length of the crane rod



- 1 Euler-Lagrange Equations
- 2 Modelling the Rotation Dynamics (Gros2012f,Gros2013b)**
- 3 Baumgarte Stabilisation (Gros2012f)
- 4 Tether Models (Pesce2003, Zanon2012, Zanon2013a)

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- 3 Matrix derivative

$$\frac{\partial}{\partial R} = \begin{bmatrix} \frac{\partial}{\partial R_{11}} & \frac{\partial}{\partial R_{12}} & \frac{\partial}{\partial R_{13}} \\ \frac{\partial}{\partial R_{21}} & \frac{\partial}{\partial R_{22}} & \frac{\partial}{\partial R_{23}} \\ \frac{\partial}{\partial R_{31}} & \frac{\partial}{\partial R_{32}} & \frac{\partial}{\partial R_{33}} \end{bmatrix}$$

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$$\frac{\partial}{\partial R} = \begin{bmatrix} \frac{\partial}{\partial R_{11}} & \frac{\partial}{\partial R_{12}} & \frac{\partial}{\partial R_{13}} \\ \frac{\partial}{\partial R_{21}} & \frac{\partial}{\partial R_{22}} & \frac{\partial}{\partial R_{23}} \\ \frac{\partial}{\partial R_{31}} & \frac{\partial}{\partial R_{32}} & \frac{\partial}{\partial R_{33}} \end{bmatrix}$$

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$$\frac{d}{dt} \frac{\partial L}{\partial \dot{R}} - \frac{\partial L}{\partial R} = F^R \quad \rightarrow \text{Weird... What is } F^R? \text{ Why } \dot{R} \text{ instead of } \omega?$$

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$$\mathcal{P}(\dot{R}) = \mathcal{U}(R^T R \Omega) = \mathcal{U}(\Omega) = \mathcal{U} \left(\begin{bmatrix} 0 & -\omega_z & \omega_y \\ \omega_z & 0 & -\omega_x \\ -\omega_y & \omega_x & 0 \end{bmatrix} \right) = \frac{1}{2} \begin{bmatrix} 2\omega_x \\ 2\omega_y \\ 2\omega_z \end{bmatrix} = \omega$$

$$L = \underbrace{T}_{T^{\text{tr}}+T^{\text{rot}}} - U - \text{tr}(\Lambda \underbrace{(R^{\text{T}}R - I)}_{C^{\text{on}}}) - \lambda^{\text{T}} C(r, R)$$

Kinetic Energy:

Orthonormality Constraint:

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Finally! **The equations of motion**

$$m\ddot{r} + \frac{\partial V + \lambda^\top C}{\partial r} = F^r$$

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$$\dot{R} = R\Omega$$

$$C(p, R) = 0$$

- 1 Euler-Lagrange Equations
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- 3 Baumgarte Stabilisation (Gros2012f)**
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From index reduction (the index-1 DAE imposes $\ddot{C} = 0$):

$$C_0 = \begin{bmatrix} C \\ \dot{C} \end{bmatrix}, \quad \dot{C}_0 = A_C C_0 = \begin{bmatrix} 0 & I \\ 0 & 0 \end{bmatrix} C_0, \quad \text{eig}(A_C) = 0$$

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$$\dot{\tilde{C}}_0 = \bar{A}_C C_0 = \begin{bmatrix} 0 & I \\ -p^2 I & -2pI \end{bmatrix} C_0, \quad \text{eig}(\bar{A}_C) = -p$$

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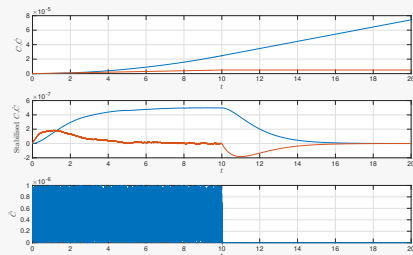
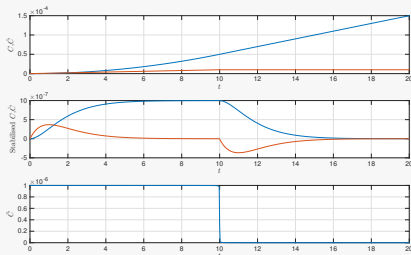


Figure: Invariant simulation with $p = 1$ and $\ddot{C} = 10^{-6}$ or $\ddot{C} = \mathcal{U}(0, 10^{-6})$

What about the orthonormality condition?

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Then:

$$\begin{aligned} \frac{d}{dt}(R^T R - I) &= R^T \dot{R} + \dot{R}^T R \\ &= R^T R \Omega + \Omega^T R^T R + \frac{\gamma}{2} R^T R \left((R^T R)^{-1} - I \right) + \frac{\gamma}{2} \left((R^T R)^{-1} - I \right) R^T R \\ &= R^T R \Omega + \Omega^T R^T R - \gamma \left(R^T R - I \right) \end{aligned}$$

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Constraints:

$$C_k = \frac{1}{2} \left((\mathbf{r}_{k,j} - \mathbf{r}_{k,j-1})^\top (\mathbf{r}_{k,j} - \mathbf{r}_{k,j-1}) - \frac{l_k}{N_k} \right)$$

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$$v_{k,j} = \sigma \dot{r}_{k,j-1} + (1 - \sigma) \dot{r}_{k,j} + w(z)$$

$$z = \sigma z_{k,j-1} + (1 - \sigma) z_{k,j}$$

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2 Drag force acting at the generalised coordinates $r_{k,j}$

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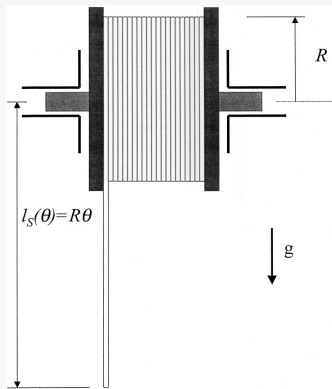
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$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) - \frac{\partial L}{\partial q} = F^q + \frac{1}{2} \frac{d}{dt} \left(\frac{\partial m}{\partial \dot{q}} v^2 \right) - \frac{1}{2} \frac{\partial m}{\partial q} v^2$$

Note that $v = v(t, q, \dot{q})$.

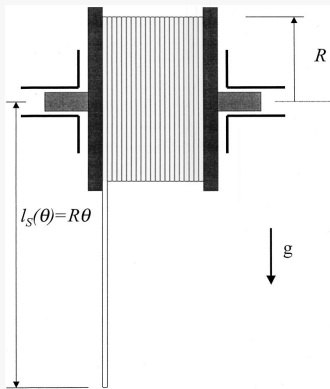
Example: Winch

Control volume: winch + suspended tether



Example: Winch

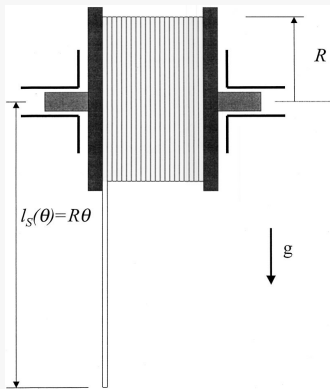
Control volume: winch + suspended tether



$$m = \mu L \quad m_s(\theta) = \mu R\theta \quad m_r(\theta) = m - \mu R\theta$$

Example: Winch

Control volume: winch + suspended tether

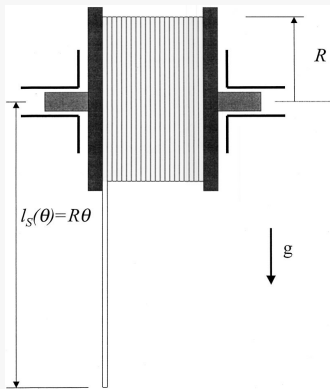


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$$T = \frac{1}{2}(J + mR^2)\dot{\theta}^2 \quad U = -\frac{1}{2}m_s(\theta)gR\theta = -\frac{1}{2}\mu gR^2\theta^2$$

Example: Winch

Control volume: winch + suspended tether



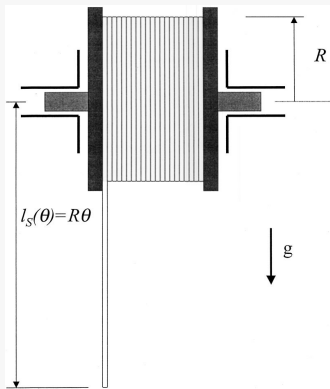
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$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) - \frac{\partial L}{\partial q} = F^q$$

Example: Winch

Control volume: winch + suspended tether



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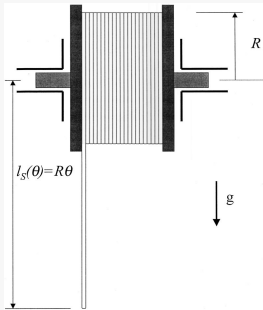
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$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) - \frac{\partial L}{\partial q} = F^q$$

$$(J + mR^2)\ddot{\theta} - \mu gR^2\theta = 0$$

Example: Winch

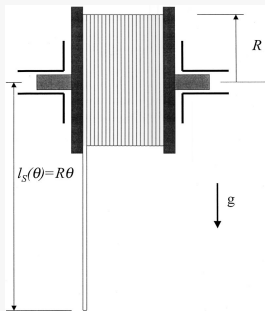
Control volume: winch



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Example: Winch

Control volume: winch

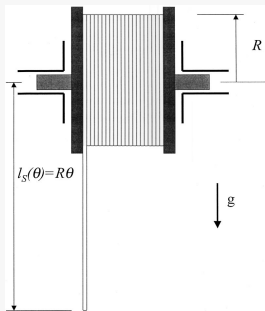


$$m = \mu L \quad m_s(\theta) = \mu R\theta \quad m_r(\theta) = m - \mu R\theta$$

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Example: Winch

Control volume: winch



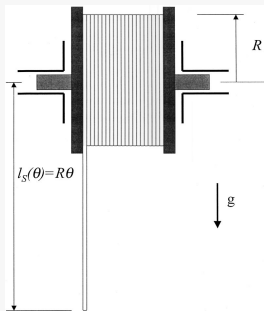
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Example: Winch

Control volume: winch



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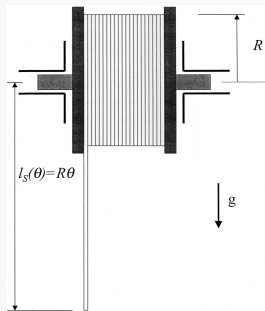
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$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) = (J + mR^2)\ddot{\theta} - \mu R^3\dot{\theta}^2 - \mu R^3\theta\ddot{\theta}$$

Example: Winch

Control volume: winch



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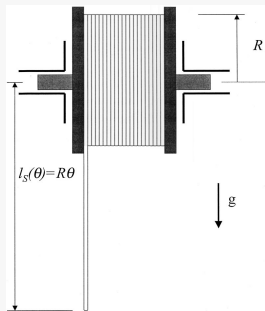
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$$\frac{\partial L}{\partial q} = -\frac{1}{2}\mu R^3\dot{\theta}^2 - \mu g R^2\theta, \quad \frac{1}{2} \frac{\partial m_r}{\partial q} v^2 = -\frac{1}{2}\mu R^3\dot{\theta}^2$$

Example: Winch

Control volume: winch



$$m = \mu L \quad m_s(\theta) = \mu R\theta \quad m_r(\theta) = m - \mu R\theta$$

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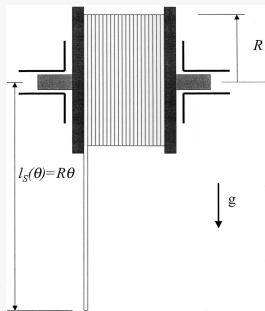
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Example: Winch

Control volume: winch



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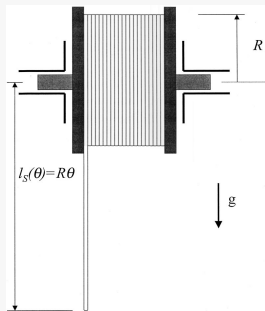
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$$F^q = (F^t + \dot{m}_r v_{m_r})R = F^t R - \mu R^3 \dot{\theta}^2$$

$$(J + mR^2)\ddot{\theta} - \frac{1}{2}\mu R^3\dot{\theta}^2 - \mu R^3\theta\ddot{\theta} - \mu g R^2\theta = F^q + \frac{1}{2}\mu R^3\dot{\theta}^2$$

Example: Winch

Control volume: winch



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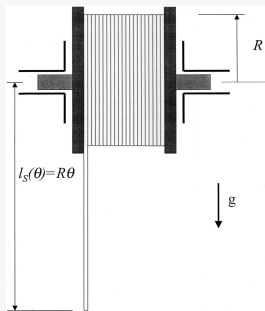
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$$\frac{d(m_s v)}{dt} - F^t - \dot{m}_s v_{m_s} = 0$$

Example: Winch

Control volume: winch



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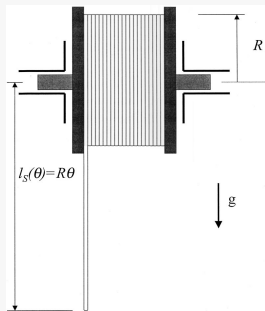
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$$\frac{d(m_s v)}{dt} - F^t - \dot{m}_s v_{m_s} = 0$$

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Example: Winch

Control volume: winch



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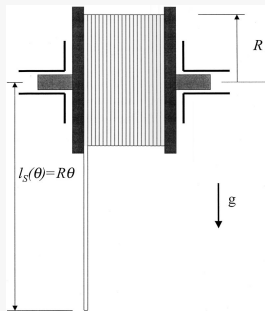
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$$\mu R^2\dot{\theta}^2 + \mu R^2\theta\ddot{\theta} - F^t - \mu R^2\dot{\theta}^2 = 0$$

$$F^t = \mu R^2\theta\ddot{\theta}$$

Example: Winch

Control volume: winch



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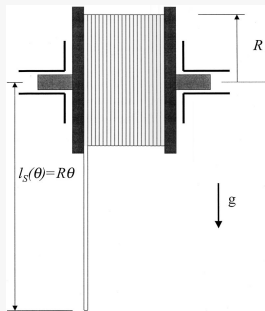
$$F^t = \mu R^2 \theta \ddot{\theta}$$

$$(J + mR^2)\ddot{\theta} - \frac{1}{2}\mu R^3 \dot{\theta}^2 - \mu R^3 \theta \ddot{\theta} - \mu g R^2 \theta = F^q + \frac{1}{2}\mu R^3 \dot{\theta}^2$$

$$(J + mR^2)\ddot{\theta} - \mu g R^2 \theta = 0$$

Example: Winch

Control volume: winch



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$$F^q = (F^t + \dot{m}_r v_{m_r})R = F^t R - \mu R^3\dot{\theta}^2$$

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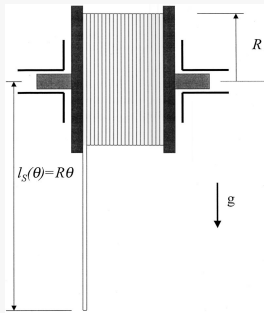
$$(J + mR^2)\ddot{\theta} - \frac{1}{2}\mu R^3\dot{\theta}^2 - \mu R^3\theta\ddot{\theta} - \mu g R^2\theta = F^q + \frac{1}{2}\mu R^3\dot{\theta}^2$$

$$(J + mR^2)\ddot{\theta} - \mu g R^2\theta = 0$$

$$(J + mR^2)\ddot{\theta} - \mu g R^2\theta - \frac{1}{2}\mu R^3\dot{\theta}^2 = 0 \quad \text{Standard EL appr.}$$

Example: Winch

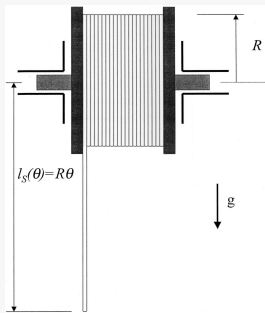
Control volume: tether



$$m = \mu L \quad m_s(\theta) = \mu R\theta \quad m_r(\theta) = m - \mu R\theta$$

Example: Winch

Control volume: tether

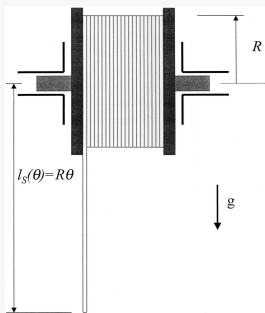


$$m = \mu L \quad m_s(\theta) = \mu R\theta \quad m_r(\theta) = m - \mu R\theta$$

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Example: Winch

Control volume: tether



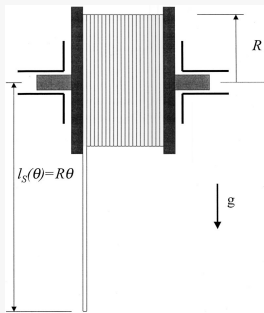
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$$\frac{\partial L}{\partial \dot{q}} = \mu R^3 \theta \dot{\theta}$$

Example: Winch

Control volume: tether



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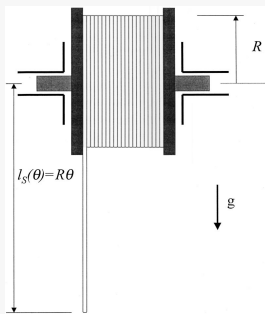
$$T_s = \frac{1}{2} \mu R^3 \theta \dot{\theta}^2 \quad U = -\frac{1}{2} \mu g R^2 \theta^2$$

$$\frac{\partial L}{\partial \dot{q}} = \mu R^3 \theta \dot{\theta}$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) = \mu R^3 \dot{\theta}^2 + \mu R^3 \theta \ddot{\theta}$$

Example: Winch

Control volume: tether



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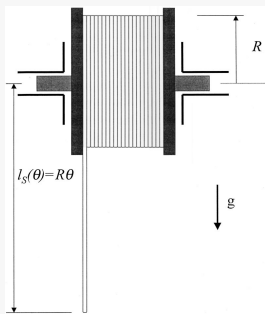
$$\frac{\partial L}{\partial \dot{q}} = \mu R^3 \theta \dot{\theta}$$

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$$\frac{\partial L}{\partial q} = \frac{1}{2} \mu R^3 \dot{\theta}^2 - \mu g R^2 \theta, \quad \frac{1}{2} \frac{\partial m_s}{\partial q} v^2 = \frac{1}{2} \mu R^3 \dot{\theta}^2$$

Example: Winch

Control volume: tether



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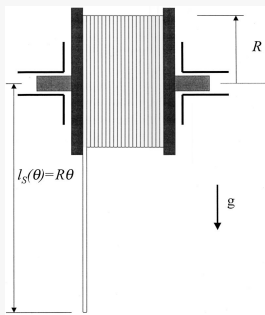
$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) = \mu R^3 \dot{\theta}^2 + \mu R^3 \theta \ddot{\theta}$$

$$\frac{\partial L}{\partial q} = \frac{1}{2} \mu R^3 \dot{\theta}^2 - \mu g R^2 \theta, \quad \frac{1}{2} \frac{\partial m_s}{\partial q} v^2 = \frac{1}{2} \mu R^3 \dot{\theta}^2$$

$$\mu R^3 \dot{\theta}^2 + \mu R^3 \theta \ddot{\theta} - \frac{1}{2} \mu R^3 \dot{\theta}^2 = F^q - \frac{1}{2} \mu R^3 \dot{\theta}^2$$

Example: Winch

Control volume: tether



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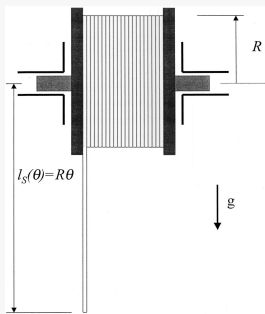
$$\frac{\partial L}{\partial q} = \frac{1}{2} \mu R^3 \dot{\theta}^2 - \mu g R^2 \theta, \quad \frac{1}{2} \frac{\partial m_s}{\partial q} v^2 = \frac{1}{2} \mu R^3 \dot{\theta}^2$$

$$\mu R^3 \dot{\theta}^2 + \mu R^3 \theta \ddot{\theta} - \frac{1}{2} \mu R^3 \dot{\theta}^2 = F^q - \frac{1}{2} \mu R^3 \dot{\theta}^2$$

$$F^q = (F^t + \dot{m}_s v_{m_s}) R = F^t R + \mu R^3 \dot{\theta}^2$$

Example: Winch

Control volume: tether



$$m = \mu L \quad m_s(\theta) = \mu R\theta \quad m_r(\theta) = m - \mu R\theta$$

$$T_s = \frac{1}{2} \mu R^3 \dot{\theta}^2 \quad U = -\frac{1}{2} \mu g R^2 \theta^2$$

$$\frac{\partial L}{\partial \dot{q}} = \mu R^3 \dot{\theta}$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) = \mu R^3 \ddot{\theta} + \mu R^3 \theta \ddot{\theta}$$

$$\frac{\partial L}{\partial q} = \frac{1}{2} \mu R^3 \dot{\theta}^2 - \mu g R^2 \theta, \quad \frac{1}{2} \frac{\partial m_s}{\partial q} v^2 = \frac{1}{2} \mu R^3 \dot{\theta}^2$$

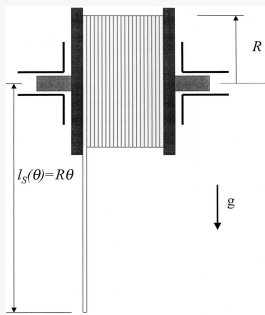
$$\mu R^3 \dot{\theta}^2 + \mu R^3 \theta \ddot{\theta} - \frac{1}{2} \mu R^3 \dot{\theta}^2 = F^q - \frac{1}{2} \mu R^3 \dot{\theta}^2$$

$$F^q = (F^t + \dot{m}_s v_{m_s}) R = F^t R + \mu R^3 \dot{\theta}^2$$

$$\frac{d(I_r \omega)}{dt} + R F^t - \dot{I}_r \omega I_r = 0$$

Example: Winch

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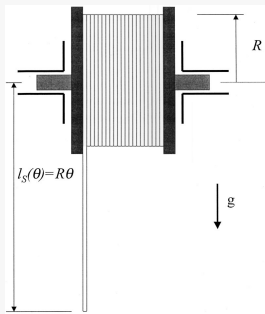
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$$-\mu R^2 \dot{\theta}^2 + (J + m R^2) \ddot{\theta} - \mu R^2 \theta \ddot{\theta} + R F^t + \mu R^2 \dot{\theta}^2 = 0$$

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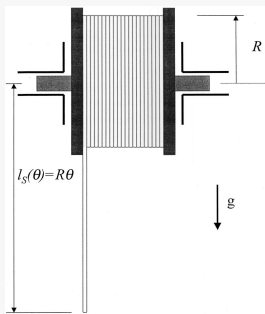
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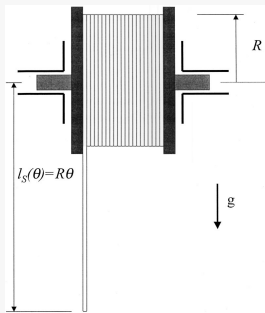
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$$(J + mR^2) \ddot{\theta} - \mu g R^2 \theta = 0$$

$$(J + mR^2) \ddot{\theta} - \mu g R^2 \theta + \frac{1}{2} \mu R^3 \dot{\theta}^2 = 0 \quad \text{Standard EL appr.}$$

$$F^q = (F^t + \dot{m}_s v_{m_s}) R = F^t R + \mu R^3 \dot{\theta}^2$$

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Exercises

- 1 Model a single kite without rotational dynamics (pointmass model) [Gros2012a, Zanon2013]
- 2 Model a single kite with rotational dynamics (but no bridle) [Gros2012f, Zanon2013c]
- 3 Model a single kite with bridle [Gros2013b]
- 4 Model a dual kite system [Zanon2014a]